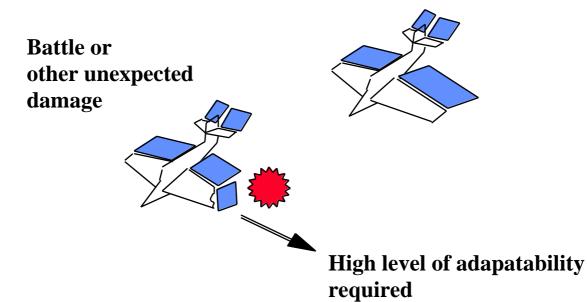
### The Grand Challenge in Flight Control

Ray Montgomery & Mike Scott
Dynamics and Control Branch
NASA Langley Research Center

May 6, 1998

#### **Best Possible Control**

Following Physical Damage During Flight



Examples exist but,

Is the technology here or imminent?

### System Requirements

- State of System Access from aero/control viewpoint
  - What control devices are still functional?
  - What is the existing surface geometry for the vehicle?
- Broadcast Alert Automatic alert of status and need
- Mission Plan Develop/modify in light of situation
  - What is best emergency mission from existing options?
    - Landing, ditching options location?
- Redesign control system to reflect mission selected
- Update Continually as mission evolves

# Examining Approaches that Promise High Adapatbility

- Adaptive Control
  - May be good for highly structured cases but, past experience is disappointing.
- Neural-Networks
  - Since there is no tie to physics of process, for reasonable behavior, not luck, the system must have seen a like event before, either in real or simulated situations, and be trained to respond properly.
- Physics-Based Schemes
  - Requires adaquate sensors and control devices for implementation
  - Current progress in computer technology offers hope of

#### **Incorporating CFD into flight control systems**

#### Available CFD theories

Viscous, compressible flow

- called Navier-Stokes theory Most complex
   Invicid, compressible theory
  - called Euler theory Still complex but needed for supersonic and transonic flow cases

Invicid, incompressible theory -

- called potential theory for fluid flow
- can be used for high subsonic with compressibility corrections added
- can be used in conjunction with a boundary layer theory (BLT) to deal with real flowphenonema

Theory of choice

# Modelling with Potential Flow

#### **Fundamental Relations:**

- Continuity Equation -> Potential Equation
- Momentum -> Bernoulli Equation for pressure distributions
- Doublet distributions over wakes -> produce circulation driven by Kutta conditions
- Modelling separation point and bubble generates *effective* surfaces to define potential flow regions

# Synopsis of Potential Theory

- Conservation of mass is fundamental equation, = 0
- V = grad() in (x,y,z) space where V is the velocity of a particle at (x,y,z) and is the velocity potential at (x,y,z)
- Requires no vorticity in flow regions considered
- To solve:
  - Green's formula -> Need only grid the near field boundaries -aircraft surfaces (with separation bubbles) and wakes
  - Panel methods well developed which incorporate far field boundary conditions analytically.
  - Time varying flow results from separation dynamics and deformation of the boundaries, e.g. control surface motions

# Synopsis of Potential Theory

#### -- Time Varying with Conservative Body Forces --

Satisfy potential equation at each in stant of time as well as new Bernoulli equation to define pressures:

$$\frac{\mathbf{w}^2}{t} + \frac{\mathbf{w}^2}{2} + \frac{p}{2} - U = f(t)$$

w here:

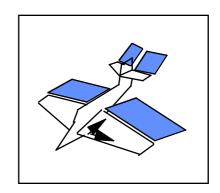
- t is time
- $\mathbf{p}$  is pres su re at (x, y, z)
- is fluid d ensity at (x, y, z)
- U is the body force potential, usually  $-\mathbf{g} \cdot \mathbf{z}$
- **f** ~ usually zero

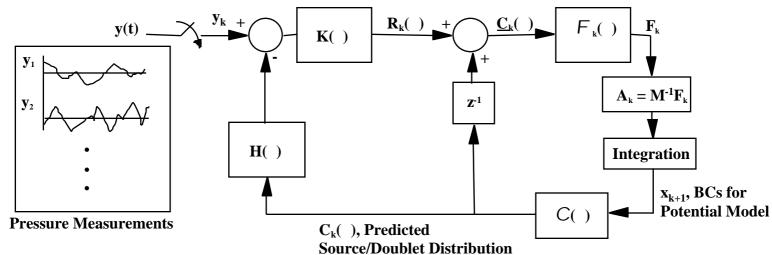
### Long Range Plan

- Refine DCB core competency in CFD
  - 2 D Panel code -> basic potential method with boundary layer
  - 3 D code for implementation
- Select target vehicle Sensor/actuator driven system
  - Need on-line surface grid generation and pressures
  - J. Foster's RPV, DARPA's X36 experiment, or other
- Architecture incorporating CFD in a flight control system
  - Study ability to predict time horizon, correlatablilty
  - Develop prediction, estimation, system ID for on-line tuning
  - Control law studies, design, integration
- Simulation and Fail-Safe Flight Demonstration
  - recover from failure in small target region of vehicle e.g. add
     vertical tail to tailless design and blow off other concepts possible.

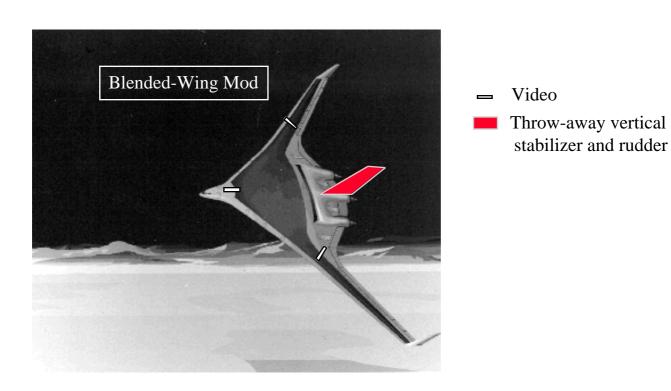
# Kalman Estimator for Real-Time Flow Control

#### **Potential Flow Model**





# Fail-Safe Flight Demonstration - Blended-Wing Model or X36 -



Now, Here's Mike#@\$@%!!

(I'll try to be quiet.)